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(54) **ROTOR YOKE AND METHOD OF MAKING THE SAME**

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(75) Inventors: **Frank B. Stamps**, Colleyville, TX (US);
Pat Tisdale, Roanoke, TX (US); **Bryan Marshall**, Mansfield, TX (US); **Tom Campbell**, Keller, TX (US)

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(73) Assignee: **Textron Innovations Inc.**, Providence, RI (US)

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Primary Examiner — Jeffrey Wollschlager

(74) *Attorney, Agent, or Firm* — James E. Walton; Brian E. Harris

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B64C 27/33 (2006.01)

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CPC **B64C 27/33** (2013.01); **B29C 70/026** (2013.01); **B29C 70/46** (2013.01); **B64C 27/32** (2013.01)

(58) **Field of Classification Search**

None

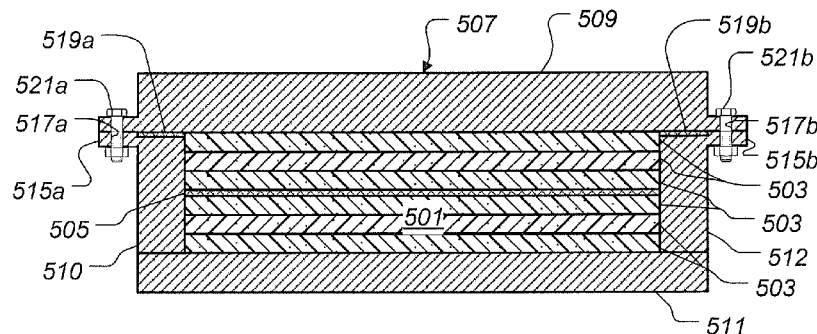
See application file for complete search history.

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ABSTRACT

A method of making a composite rotor yoke includes preparing a molded rotor yoke in a closed cavity tool and possibly machining at least one portion of the molded rotor yoke to form the rotor yoke. In the preferred embodiment, preparing the molded rotor yoke is accomplished by applying layers of uncured low-flow composite material and a layer of uncured high-flow adhesive; substantially enclosing the uncured molded rotor yoke in the closed cavity tool; applying pressure so as to compress the uncured molded rotor yoke; and curing the uncured molded rotor yoke. During the curing process, the high-flow adhesive bleeds out of the molded rotor yoke, thereby preventing marclers from forming through movement of the low-flow composite material.

13 Claims, 5 Drawing Sheets



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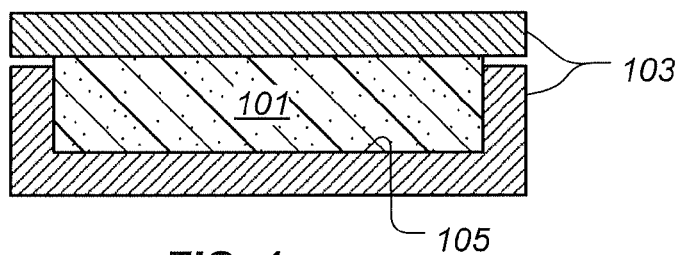


FIG. 1
(PRIOR ART)

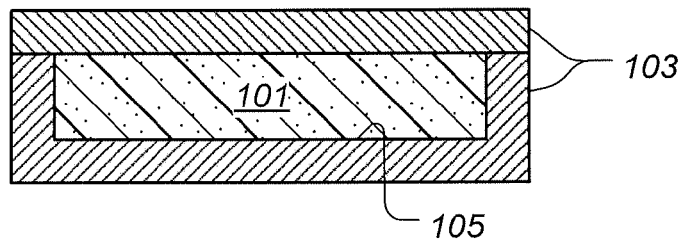


FIG. 2
(PRIOR ART)

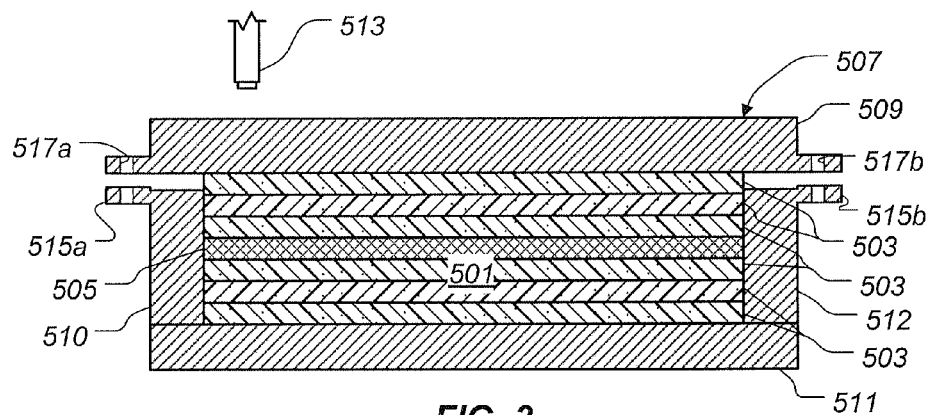


FIG. 3

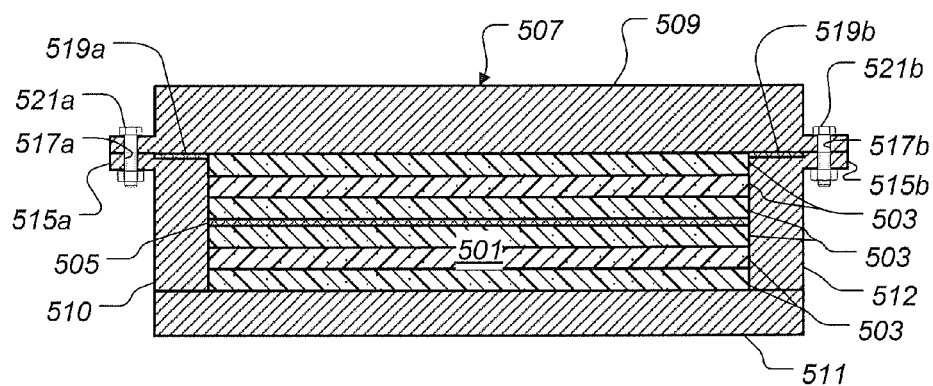


FIG. 4

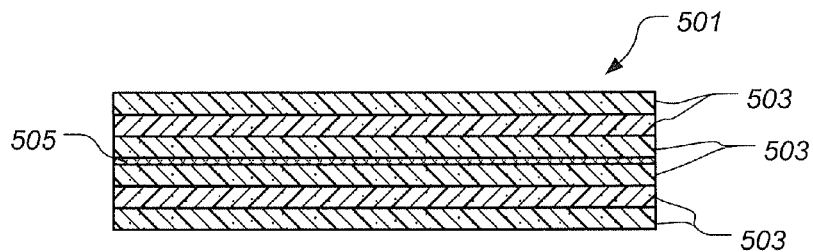
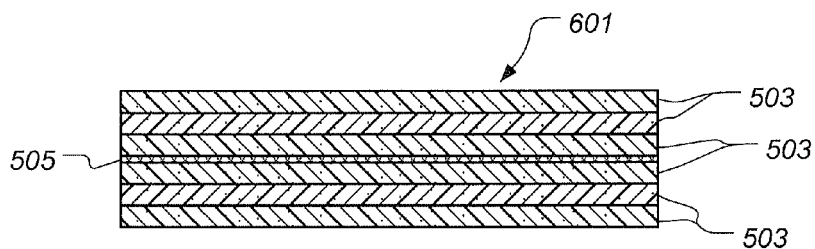
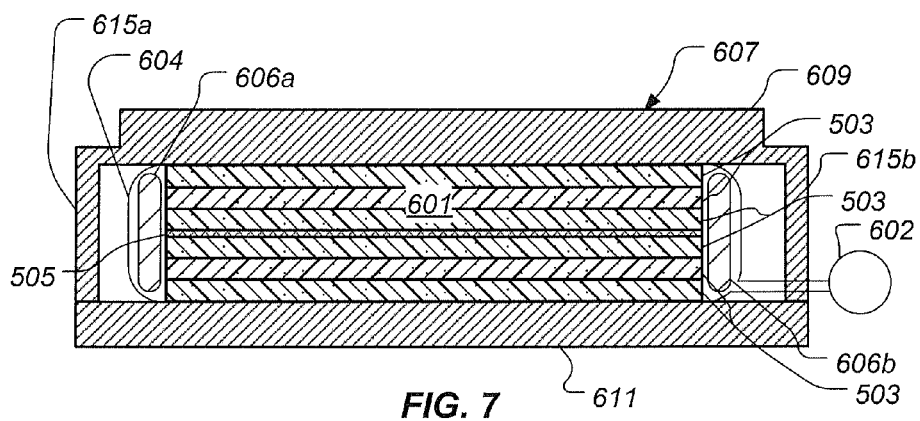
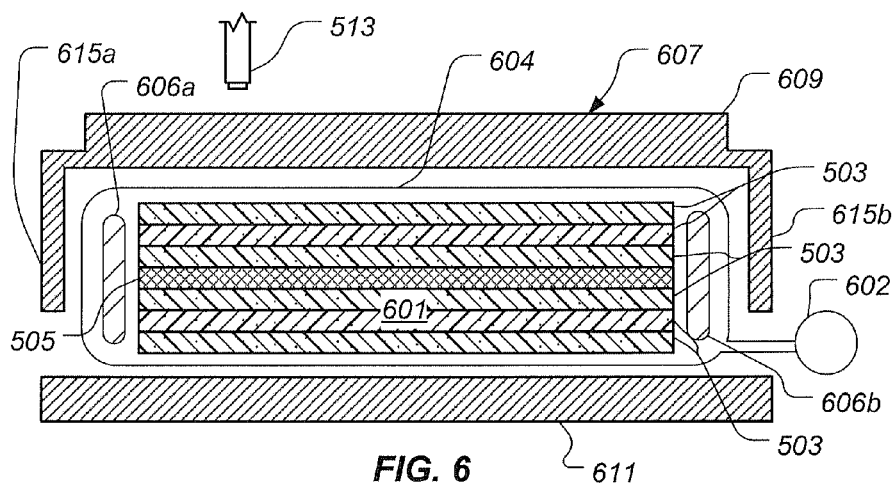


FIG. 5



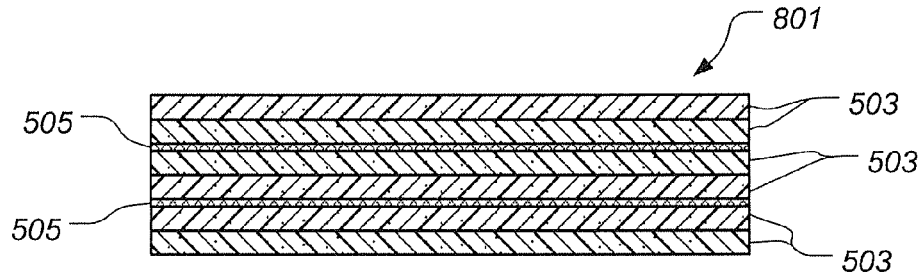


FIG. 9

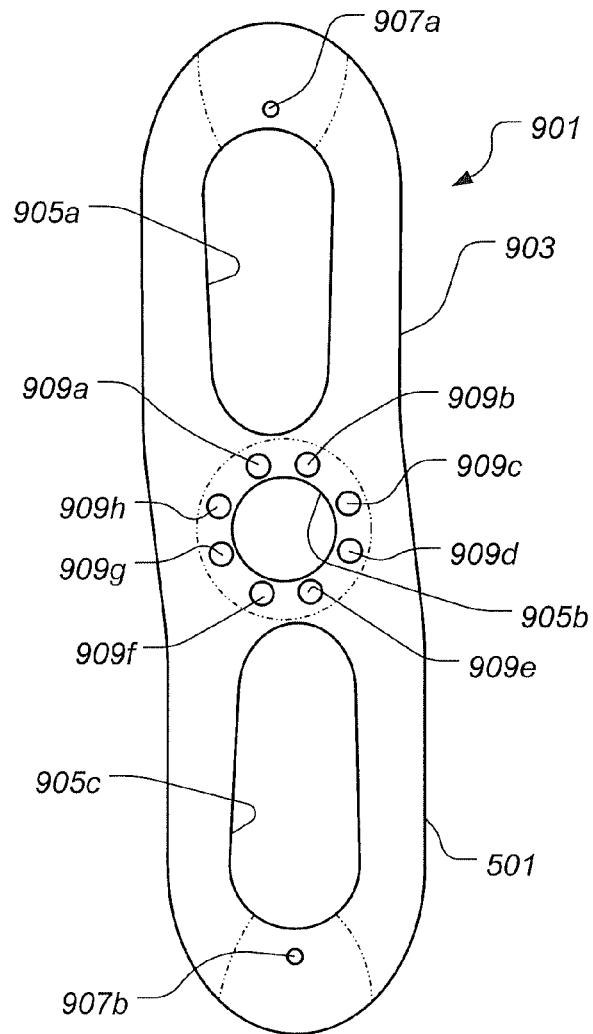


FIG. 10

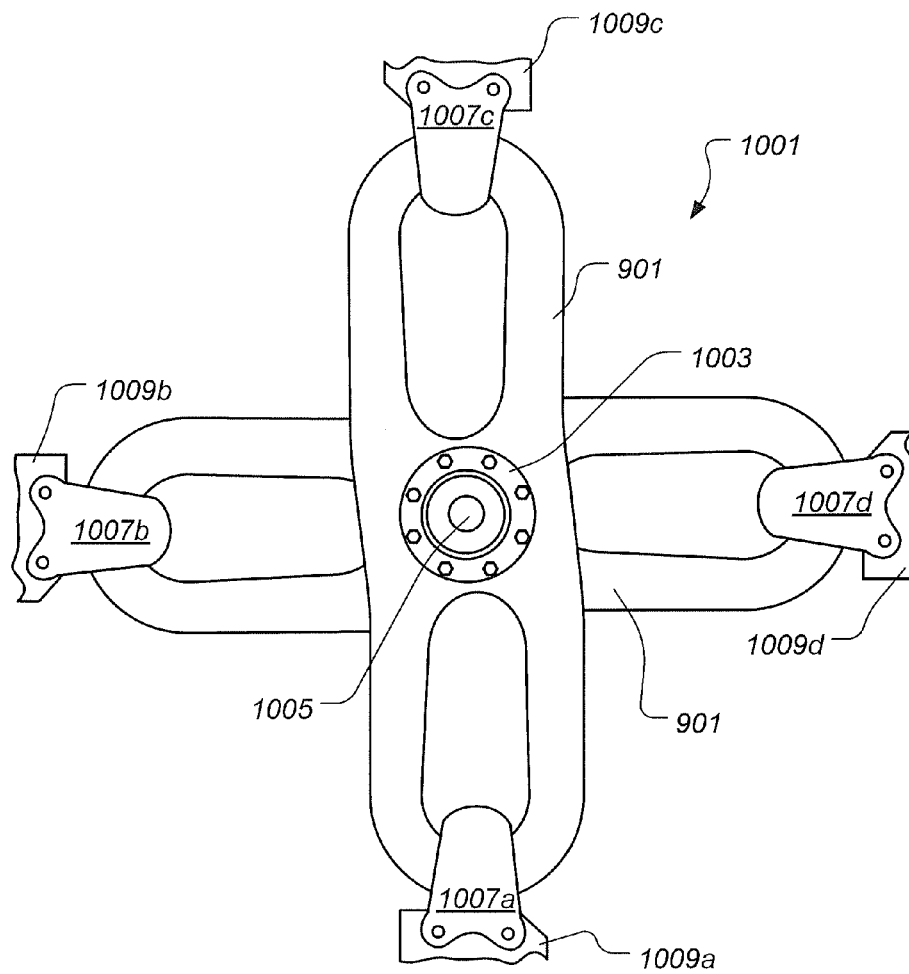


FIG. 11

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ROTOR YOKE AND METHOD OF MAKING THE SAME

TECHNICAL FIELD

The present application relates to rotorcraft and, in particular, to yokes for coupling helicopter blades to a mast.

DESCRIPTION OF THE PRIOR ART

Each blade of the main rotor assembly of a rotorcraft must be connected to a main support mast, usually by means of a rotor yoke, in a manner allowing several degrees of freedom. Such an interconnection is subjected to high and repeated stresses of both torsional and centrifugal natures, and is therefore an extremely important component of the aircraft. Each blade must be able to rotate about its longitudinal axis to provide pitch control. Each blade must be able to flap in a direction perpendicular to the rotor plane to accommodate vertical loads. In some instances, each blade must be able to pivot within the rotor plane to provide for lead-lag control. The manner in which the blades are secured to the main support mast enables a rotorcraft to be controlled and maneuvered in flight.

Various types of rotor yokes have been utilized to interconnect the rotorcraft blades and the support mast. Metal rotor yokes have suffered from the disadvantages of weight, cost, high maintenance requirements, and low useful life. There have been several attempts to eliminate one or more of the articulations in such couplings in order to simplify construction and reduce costs. Some rotor yokes are pivotally secured to the support mast, and are characterized by a flat plate construction resilient enough to act as a virtual hinge and thereby accommodate flapping of the blades.

More recently, glass fibers and other composite materials have been employed in the fabrication of rotorcraft rotor system components. In comparison to a machined metal forging, glass fibers and other composite materials have more favorable fatigue characteristics resulting in longer useful life. In addition, the use of such materials simplifies construction and reduces costs. Referring to FIGS. 1 and 2, composite rotor yokes, such as a rotor yoke 101 are conventionally cured in a rigid, closed mold, such as mold 103, to form the overall shape of the rotor yoke. One of the problems encountered concerning such rotorcraft rotor yokes has been distortion or "marcelling" of the fibers in the rotor yoke during the curing process. Because the uncured rotor yoke is forced to conform to the cavity, such as cavity 105 formed by the closed mold, mechanical stresses can be induced in the uncured rotor yoke. The fibers are substantially unconstrained during certain portions of the curing cycle when the resin matrix in which the fibers are disposed is in a semi-liquid or liquid state. The induced stress in the uncured rotor yoke is relieved via movement or distortion of the fibers within the resin matrix. The fibers can be captured in their distorted or marcelled state when the resin crosslinks in thermosetting composite materials or when the resin is cooled in thermoplastic composite materials.

There are many designs of rotorcraft yokes well known in the art; however, considerable shortcomings remain.

BRIEF DESCRIPTION OF THE DRAWINGS

The novel features believed characteristic of the present application are set forth in the appended claims. However, the system itself, as well as a preferred mode of use, and further objectives and advantages thereof, will best be understood by

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reference to the following detailed description when read in conjunction with the accompanying drawings, in which the leftmost significant digit(s) in the reference numerals denote(s) the first figure in which the respective reference numerals appear, wherein:

FIGS. 1 and 2 are stylized, cross-sectional views illustrating a conventional method for manufacturing a composite rotor yoke for a rotorcraft according to prior art;

FIGS. 3-5 are stylized, cross-sectional views depicting the method of manufacturing a composite molded rotor yoke according to the preferred embodiment of the present application;

FIGS. 6-8 are stylized, cross-sectional views depicting the method of manufacturing a composite molded rotor yoke according to an alternative embodiment of the present application;

FIG. 9 is a stylized, cross-sectional view of a molded rotor yoke according to an alternative embodiment of the present application;

FIG. 10 is a top, plan view of a composite rotor yoke according to the preferred embodiment of the present application; and

FIG. 11 is a top, plan view of a rotor hub incorporating a pair of composite rotor yokes of FIG. 10 according to the preferred embodiment of the present application.

While the system of the present application is susceptible to various modifications and alternative forms, specific embodiments thereof have been shown by way of example in the drawings and are herein described in detail. It should be understood, however, that the description herein of specific embodiments is not intended to limit the present application to the particular forms disclosed, but on the contrary, the intention is to cover all modifications, equivalents, and alternatives falling within the spirit and scope of the present application as defined by the appended claims.

DESCRIPTION OF THE PREFERRED EMBODIMENT

The system of the present application represents a composite rotor yoke for a rotorcraft produced using a closed cavity curing tool. Preferably, the composite rotor yoke is laid-up using an automated fiber placement process, but may also be laid-up by hand. After curing, the cured composite rotor yoke may be machined to add any desired features.

Illustrative embodiments of the present application are described below. In the interest of clarity, not all features of an actual implementation are described in this specification. It will of course be appreciated that in the development of any such actual embodiment, numerous implementation-specific decisions must be made to achieve the developer's specific goals, such as compliance with system-related and business-related constraints, which will vary from one implementation to another. Moreover, it will be appreciated that such a development effort might be complex and time-consuming but would nevertheless be a routine undertaking for those of ordinary skill in the art having the benefit of this disclosure.

As used herein, reference may be made to the spatial relationships between various components and to the spatial orientation of various aspects of components as the devices are depicted in the attached drawings. However, as will be recognized by those skilled in the art after a complete reading of the present application, the devices, members, apparatuses, etc. described herein may be positioned in any desired orientation. Thus, the use of terms such as "above," "below," "upper," "lower," or other like terms to describe a spatial relationship between various components or to describe the

spatial orientation of aspects of such components should be understood to describe a relative relationship between the components or a spatial orientation of aspects of such components, respectively, as the device described herein may be oriented in any desired direction.

Referring to FIGS. 3-5 in the drawings, the preferred embodiment of a composite molded rotor yoke **501** of a rotor yoke **901** (shown in FIG. 10) is fabricated by applying a plurality of layers of an uncured low-flow composite material **503** and a layer of a high-flow adhesive **505** into a closed cavity tool **507**. Note that the geometric configuration of tool **507** is merely exemplary of the widely diverse geometric configurations of closed cavity tools contemplated by the present application. It should be appreciated that tools **509**, **510**, **511**, and **512** may be split into multiple tools, or combined to form a fewer number of tool parts. For example, side tools **510** and **512** may be integrated into a second tool **511** so as to form one integral tool part. Closed cavity tool **507** preferably has a rigid first tool **509**, a rigid second tool **511**, and rigid side tools **510** and **512**. However, closed cavity tool **507** may also have a semi-rigid first tool **509**, a semi-rigid second tool **511**, semi-rigid side tools **510** and **512**, or any combination of semi-rigid and rigid tools **509**, **511**, **510**, and **512**. Closed cavity tool **507** provides tolerance and contour control by conforming the primary surfaces of molded rotor yoke **501** to the inside surfaces of closed cavity tool **507**. Closed cavity tool **507** may also be referred to as a “two-sided” tool or other language describing that tool **507** substantially encloses molded rotor yoke **501**.

Uncured, composite molded rotor yoke **501** is formed when the desired number of layers, also referred to as “plies”, of low-flow composite material **503** and high-flow adhesive **505** have been applied into tool **507**, in the desired geometry. Layers of low-flow composite material may be laid by hand, or by fiber placement machine **513**. High-flow adhesive **505** may also be laid by hand, or by a machine similar to fiber placement machine **513**. Closed cavity tool **507** is capable of compressing molded rotor yoke **501** into a desired thickness and geometry. Closed cavity tool **507** may include stops **515a** and **515b** so that first tool **509** and second tool **511** of tool **507** will stop the compression of molded rotor yoke **501** at the desired thickness of molded rotor yoke **501**. Closed cavity tool **507** may also include fastener holes **517a** and **517b** so that closed cavity tool **507** could be fastened closed with fasteners **521a** and **521b** at any time during or after the curing process.

Uncured low-flow composite material **503** preferably includes glass fibers disposed in an uncured epoxy, in the form of a prepreg, although the present application contemplates other materials for molded rotor yoke **501**. An example of low-flow composite material **503** is HexPly 8552 made by Hexcel Composites. For this application, the term “prepreg” is typically sheets of fibers impregnated in uncured epoxy or adhesive. Uncured low-flow composite material **503** can then be cut to size and laid into tool **507**, either by hand or with fiber placement machine **513**. Low-flow composite material **503**, if used in the absence of high-flow adhesive **505**, could be any composite material that has such a high viscosity so as to possibly produce marcells when subjected to a curing process inside a closed cavity tool **507**. “Marcells” are wrinkles in the fiber structure of a composite that severely compromise the structural integrity of a composite part. Marcells are often created when low-flow composite material **503**, if used in the absence of high-flow adhesive **505**, is forced to conform to the inside surface of closed cavity tool **507**, causing the highly viscous epoxy to flow and distort fibers. Complete curing of

the distorted fibers, or marcells, causes the fibers to be permanently fixed in the marcelled state.

The system of the present application seeks to at least prevent the formation of “marcells” by applying layers of uncured low-flow composite material **503**, as well as a layer of uncured high-flow adhesive **505**, in a configuration to produce molded rotor yoke **501**. In an alternative embodiment, a plurality of layers of high-flow adhesive **505** may also be used in conjunction with a plurality of layers of low-flow composite material **503**, to form molded rotor yoke **801** (see FIG. 9). The use of uncured high-flow adhesive **505** in conjunction with low-flow composite material **503** provides a material with low viscous properties that easily bleeds out of tool **507** during curing, thereby preventing low-flow composite material **503** from forming marcells. An example of high-flow adhesive **505** is AF163 made by 3M. High-flow adhesive **505** is preferably, in its uncured state, a film, but may also be a paste. High-flow adhesive **505** may also be impregnated with fibers or scrim in order to tailor strength and final thickness control.

FIG. 4 depicts molded rotor yoke **501** during the curing process. Closed cavity tool **507** has been compressed in order to rid the part of any voids, or air bubbles, and to conform molded rotor yoke **501** to the contours of two sided tool **507**. Mechanical stops **515a** and **515b** of second tool **511** are shown in contact with first tool **509**. Contact between stops **515a** and **515b**, and first tool **509**, signify the desired final thickness of molded rotor yoke **501** has been reached. Bleed outs **519a** and **519b** of high-flow adhesive **505** are caused from compression of tool **507**. During curing, high-flow adhesive **505** may either partially or completely bleed out from molded rotor yokes **501** and **801**, depending on parameters controlling compression of tool **507**. Because high-flow adhesive **505** has a lower viscosity than that of low-flow composite material **503**, high-flow adhesive **505** bleeds out of tool **507** instead of low-flow composite material **503**. In some embodiments, the viscosity difference between high-flow adhesive **505** and low-flow composite material **503** may not be large, which could result in bleed outs **519a** and **519b** being a combination of high-flow adhesive **505** and low-flow composite material **503**. Curing the uncured molded rotor yoke **501** is accomplished by applying at least one of heat, pressure, and time. In alternative embodiment, the curing of rotor yoke molded rotor yoke **501** may involve subjecting molded rotor yoke **501** to a vacuum. Another alternative embodiment involves releasably coupling first tool **509** and second tool **511**, via fasteners **521a** and **521b**, while allowed uncured molded rotor yoke **501** to soak at an ambient temperature environment. The specific amount of variables such as heat, pressure, or time, depend up on at least the specific curing requirements of the low-flow composite material **503** and high-flow adhesive **505** used to form molded rotor yoke **501**. The location of bleed outs **519a** and **519b** may be located anywhere on tool **507** that allows high-flow adhesive **505** to escape or bleed out during the curing process; however, it is preferred that bleed outs **519a** and **519b** be located on the upper side of tool **507** so as to prevent the introduction of air bubbles into molded rotor yoke **501**.

FIG. 5 depicts molded rotor yoke **501** after the curing cycle has been completed. Bleed outs **519a** and **519b** have been removed from molded rotor yoke **501** by force and then lightly sanded to remove any sharp edges. Layer of high-flow adhesive **505** has decreased in thickness due to partially bleeding out during the curing process.

Referring to FIGS. 6-8 in the drawings, an alternative embodiment of a composite molded rotor yoke **601** of rotor yoke **901** (shown in FIG. 10) is fabricated by applying a

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plurality of layers of uncured low-flow composite material **503** and a layer of high-flow adhesive **505** into a closed cavity tool **607**. Note that the geometric configuration of tool **607** is merely exemplary of the widely diverse geometric configurations of closed cavity tools contemplated by the present application. Closed cavity tool **607** preferably has a rigid first tool **609** and a rigid second tool **611**. However, closed cavity tool **607** may also have a semi-rigid first tool **609**, a semi-rigid second tool **611**, or any combination of semi-rigid and rigid tools **609** and **611**. It should be appreciated that tools **609** and **611** may be split into multiple tools. For example, first tool **609** may be formed as two separate tools. Closed cavity tool **607** provides tolerance and contour control by conforming the primary surfaces of molded rotor yoke **601** to the inside surfaces of closed cavity tool **607**. Closed cavity tool **607** may also be referred to as a “two-sided” tool or other language describing that tool **607** substantially encloses molded rotor yoke **601**.

Uncured, composite molded rotor yoke **601** is formed when the desired number of layers, also referred to as “plies”, of low-flow composite material **503** and high-flow adhesive **505** have been applied into tool **607**, in the desired geometry. Layers of low-flow composite material may be laid by hand, or by fiber placement machine **513**. High-flow adhesive **505** may also be laid by hand, or by a machine similar to fiber placement machine **513**. Closed cavity tool **607** is capable of compressing molded rotor yoke **601** into a desired thickness and geometry. Closed cavity tool **607** may include stops **615a** and **615b** so that first tool **609** and second tool **611** of tool **607** will stop the compression of molded rotor yoke **601** at the desired thickness of molded rotor yoke **601**. It should be appreciated that stops **615a** and **615b** may be alternatively integrated into second tool **611**, or stops **615a** and **615b** may be separate parts while remaining configured to stop the compression of first tool **609** and second tool **611** at the appropriate desired thickness of molded rotor yoke **601**.

Uncured low-flow composite material **503** preferably includes glass fibers disposed in an uncured epoxy, in the form of a prepreg, although the present application contemplates other materials for molded rotor yoke **601**. An example of low-flow composite material **503** is HexPly 8552 made by Hexcel Composites. For this application, the term “prepreg” is typically sheets of fibers impregnated in uncured epoxy or adhesive. Uncured low-flow composite material **503** can then be cut to size and laid into tool **607**, either by hand or with fiber placement machine **513**. Low-flow composite material **503**, if used in the absence of high-flow adhesive **505**, could be any composite material that has such a high viscosity so as to possibly produce marcells when subjected to a curing process inside a closed cavity tool **607**. “Marcells” are wrinkles in the fiber structure of a composite that severely compromise the structural integrity of a composite part. Marcells are often created when low-flow composite material **503**, if used in the absence of high-flow adhesive **505**, is forced to conform to inside surfaces of closed cavity tool **607**, causing the highly viscous epoxy to flow and distort fibers. Complete curing of the distorted fibers, or marcells, causes the fibers to be permanently fixed in the marcelled state.

Closed cavity tool **607** preferably does not have side tools or bleed out portions; instead, tool **607** has space around the periphery of molded rotor yoke **601** for an absorbent material **606a** and **606b**. Absorbent material **606a** and **606b** functions at least to absorb high-flow adhesive **505** as it bleeds out of molded rotor yoke **601** during the curing process, as further explained below. Additionally a vacuum bag **604** is configured in order for a vacuum pump **602** to draw a vacuum on molded rotor yoke **601** during the curing process. The draw-

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ing of a vacuum on molded rotor yoke **601** during the curing process acts to help remove air bubbles from molded rotor yoke **601**. Though vacuum bag **604** is shown as encapsulating only molded rotor yoke **601**, it should be appreciated that in some embodiments vacuum bag **604** may also surround either first tool **609**, second tool **611**, or both first tool **609** and second tool **611**. It should be appreciated that other materials, such as a breather material, may be used in conjunction with vacuum bag **604** in order to draw a vacuum on molded rotor yoke **601**.

The system of the present application seeks to at least prevent the formation of “marcells” by applying layers of uncured low-flow composite material **503**, as well as a layer of uncured high-flow adhesive **505**, in a configuration to produce molded rotor yoke **601**. In an alternative embodiment, a plurality of layers of high-flow adhesive **505** may also be used in conjunction with a plurality of layers of low-flow composite material **503**, to form molded rotor yoke **801** (see FIG. 9). The use of uncured high-flow adhesive **505** in conjunction with low-flow composite material **503** provides a material with low viscous properties that easily bleeds out of molded rotor yoke **601** during curing, thereby preventing the formation of marcells. An example of high-flow adhesive **505** is AF163 made by 3M. High-flow adhesive **505** is preferably, in its uncured state, a film, but may also be a paste. High-flow adhesive **505** may also be impregnated with fibers or scrim in order to tailor strength and final thickness control.

FIG. 7 depicts molded rotor yoke **601** during the curing process. Closed cavity tool **607** has been compressed in order to rid the part of any voids, or air bubbles, and to conform molded rotor yoke **601** to the primary contours of two sided tool **607**. Mechanical stops **615a** and **615b** of first tool **609** are shown in contact with second tool **611**. Contact between stops **615a** and **615b**, and second tool **611**, signify the desired final thickness of molded rotor yoke **601** has been reached. During this process, high-flow adhesive **505** bleeds into absorbent material **606a** and **606b** due to compression of tool **607** and the drawing of a vacuum through vacuum pump **602**. During curing, high-flow adhesive **505** may either partially or completely bleed out from molded rotor yokes **601** and **801**, depending on the parameters controlling compression of tool **607**. Because high-flow adhesive **505** has a lower viscosity than that of low-flow composite material **503**, high-flow adhesive **505** bleeds out of tool **607** instead of low-flow composite material **503**. In some embodiments, the viscosity difference between high-flow adhesive **505** and low-flow composite material **503** may not be large, which could result in a small amount of resin from low-flow composite material **503** with the high-flow adhesive **505** bleeding into absorbent material **606a** and **606b**. Curing the uncured molded rotor yoke **601** is accomplished by applying at least one of heat, pressure, and time. An alternative embodiment involves releasably coupling first tool **609** and second tool **611**, via fasteners while allowed uncured molded rotor yoke **601** to soak at an ambient temperature environment. The specific amount of variables such as heat, pressure, or time, depend on at least the specific curing requirements of the low-flow composite material **503** and high-flow adhesive **505** used to form molded rotor yoke **601**.

FIG. 8 depicts molded rotor yoke **601** after the curing cycle has been completed. The layer of high-flow adhesive **505** has decreased in thickness because of adhesive **505** partially bleeding out during the curing process.

FIG. 9 depicts an alternative embodiment of molded rotor yoke **801** that was formed by either the process used to form molded rotor yoke **501** or molded rotor yoke **601**, except that a plurality of layers of high-flow adhesive **505** were used.

Multiple layers of high-flow adhesive **505** allows for a greater amount of high-flow adhesive **505** available for bleed out. Using a plurality of layers of high-flow adhesive **505** can be advantageous for many reasons, such as for tailoring the material properties of molded rotor yoke **801**, as well as for allowing for greater contour variation in tools **507** and **607**.

FIG. **10** depicts a preferred embodiment of a rotorcraft rotor yoke **901**. Rotor yoke **901** is preferably made from molded rotor yoke **501**, but may also be made from molded rotor yoke **601** or **801**. Rotor yoke **901** exemplifies features that may be machined into molded rotor yokes **501**, **601**, and **801** using a machining tool such as a five-axis machine tool. Such features include: inner openings **905a**, **905b**, and **905c**, outer periphery **903**, rotor blade attachment holes **907a** and **907b**, and mast attachment holes **909a-909h**, to name a few. Rotor yoke **901** is formed when all machining operations are complete. It should be noted that, in one embodiment, molded rotor yoke **501**, **601**, and **801** are oversized, i.e., is larger in physical dimensions than rotor yoke **901** to provide material for removal during a machining process to form machine outer periphery **903** (shown in FIG. **10**). Moreover, dimensions of openings **905a**, **905b**, and **905c** maybe incorporated into molded rotor yokes **501**, **601**, or **801** and then increased in size by a subsequent machining operation.

FIG. **11** depicts the preferred embodiment of a rotor hub **1001** comprising a pair of rotor yokes **901**, a coupling **1003** that mechanically couples rotor yokes **901** to a mast **1005**, and fittings **1007a-1007d** that mechanically couple rotor yokes **901** to rotor blades **1009a-1009d**, respectively. The configuration of rotor hub **1001** depicted in FIG. **11** is merely exemplary of the widely various embodiments of rotor hub **1001** contemplated by the present application.

It should be noted, that the methods disclosed in the present application can be applied to form composite structures other than rotor yoke **901** or molded rotor yokes **501**, **601**, and **801**. Any composite structure that uses low-flow composite material **503** that is susceptible to marcelling, or other fiber distortion, during compression in a closed cavity tool such as closed cavity tools **507** or **607**, would benefit from the methods disclosed in the present application. Examples of composite structures that would fall into this category include, but not limited to, composite structures used in aircraft, wind turbines, automobiles, marine vehicles, etc.

The system of the present application provides significant advantages, including: (1) ability to use two-side tooling while providing a composite rotor yoke without marcelling of the reinforcing fibers thereof; (2) enabling the use of toughened, low-flow resins in composite rotor yokes in a closed cavity tool; and (3) providing a composite rotor yoke that requires less composite fabrication time, cost, and effort with less part-to-part variation than conventional rotor yokes.

The particular embodiments of the system of the present application disclosed may be modified and practiced in different but equivalent manners apparent to those skilled in the art having the benefit of the teachings herein. Furthermore, no limitations are intended to the details of construction or design herein shown, other than as described in the claims below. It is therefore evident that the particular embodiments disclosed above may be altered or modified and all such variations are considered within the scope and spirit of the present application. Accordingly, the protection sought herein is as set forth in the claims below. It is apparent that a system with significant advantages has been described and illustrated. Although the system of the present application is shown in a limited number of forms, it is not limited to just

these forms, but is amenable to various changes and modifications without departing from the spirit thereof.

The invention claimed is:

1. A method of making a rotor yoke, comprising:
 - preparing a molded rotor yoke in a closed cavity tool, comprising:
 - applying layers of uncured low-flow composite material, the low-flow composite material comprising fibers impregnated in uncured epoxy or adhesive, and a layer of uncured high-flow adhesive, the layer of high-flow adhesive lacking fibers impregnated therein, the layers being applied in a configuration to produce an uncured molded rotor yoke;
 - substantially enclosing the uncured molded rotor yoke in the closed cavity tool, wherein the closed cavity tool comprises a first tool and a second tool;
 - forcing the first tool and the second tool together, so as to compress the uncured molded rotor yoke; and
 - curing the uncured molded rotor yoke.
 2. The method according to claim 1, further comprising: machining at least one portion of the molded rotor yoke to form the rotor yoke.
 3. The method according to claim 1, wherein the curing of the uncured molded rotor yoke includes applying at least one of heat, pressure, and time.
 4. The method according to claim 3, wherein the curing of the uncured molded rotor yoke further includes releasably coupling the first tool and the second tool together while allowing the molded rotor yoke to soak at an ambient temperature environment.
 5. The method according to claim 1, wherein the applying of the layers of an uncured low-flow composite material and the layer of uncured high-flow adhesive includes an automated fiber placement machine.
 6. The method according to claim 1, wherein the forcing of the first tool and the second tool together so as to compress the uncured molded rotor yoke forces at least a portion of the uncured high-flow adhesive to bleed out.
 7. The method according to claim 1, wherein the uncured low-flow composite material comprises a plurality of glass fibers.
 8. The method according to claim 2, wherein the machining of at least one portion of the molded rotor yoke to form the rotor yoke includes machining the cured molded rotor yoke to form mast attachment holes.
 9. The method according to claim 1, wherein the preparing of the molded rotor yoke further comprises:
 - drawing a vacuum in a vacuum bag enclosed around the molded rotor yoke.
 10. The method according to claim 1, wherein at least one of the first tool and second tool comprises multiple tools.
 11. The method according to claim 1, wherein the closed cavity tool further comprises:
 - at least one side tool adjacent to a periphery of the molded rotor yoke.
 12. The method according to claim 1, wherein the closed cavity tool further comprises at least one tool stop for controlling a thickness of the molded rotor yoke.
 13. The method according to claim 1, wherein the preparing of the molded rotor yoke further comprises:
 - applying an absorbent material around a periphery of the uncured molded rotor yoke for absorbing any bleed out of the high-flow adhesive.